THIS DOCUMENT MUST BE KEPT IN THE AIRPLANE AT ALL TIMES

CESSNA AIRCRAFT COMPANY

WICHITA, KANSAS.

CAA IDENTIFICATION

C.A.A. APPROVED APPROVAL BASIS
LANDPLANE CAR 3
SKIPLANE CAR 4
17 JUNE 1949
UTILITY CATEGORY

MODEL 140A (C-90)

AIRPLANE FLIGHT MANUAL

LIMITATIONS LANDPLANE OR SKIPLANE

THE FOLLOWING LIMITATIONS ARE TO BE OBSERVED IN THE OPERATION OF THIS AIRPLANE EQUIPPED WITH A CONTINENTAL C90-14F ENGINE.

ENGINE LIMITS

FOR ALL OPERATIONS 2475 R.P.M. (90 H.P.)

FUEL SYSTEM

80 MINIMUM OCTANE AVIATION GASOLINE. (USABLE CAPACITY TWO TANKS - 10 1/2 GALLONS EACH -- 21 TOTAL).

PROPELLER

- 1. WOOD FIXED PITCH STATIC R.P.M. AT MAXIMUM PERMISSIBLE THROTTLE SETTING -HOT HORE THAN 2160 R.P.M., HOT LESS THAN 1900 R.P.M. *
 (DIAMETER MAXIMUM 74 INCHES; MINIMUM 70.5 INCHES.)
 2. MCCAULEY 1890 STATIC R.P.M. AT MAXIMUM PERMISSIBLE THROTTLE SETTING -HOT HORE THAN 2350 R.P.M., HOT LESS THAN 2000 R.P.M. *
 (DIAMETER MAXIMUM 71 INCHES; MINIMUM 69.5 INCHES.)
 *NO ADDITIONAL STATIC R.P.M. TOLERANCE PERMITTED.

ENGINE INSTRUMENTS

- STRUMENTS

 1. OIL TEMPERATURE NORMAL OPERATING RANGE 100°-225° (GREEN ARC)

 UNSAFE ABOVE 225° (RED LINE)

 2. OIL PRESSURE NORMAL RANGE 30-40 PSI (GREEN ARC). IDLING PRESSURE 5 PSI MIN.

 MAXIMUM PRESSURE 50 PSI (RED LINE).

 3. TACHOMETER DO NOT EXCEED RATED ENGINE SPEED 2475 R.P.M. (RED LINE).

 NORMAL OPERATING RANGE 2000-2350 R.P.M. (GREEN ARC).

 4. FUEL QUANTITY GAUGE DO NOT TAKE-OFF ON A FUEL TANK LESS THAN 1/4 FULL.

 APPROXIMATELY 2 GALLONS OF FUEL UNUSABLE IN NORMAL FLIGHT MANEUVERS REMAINS
 IN EACH TANK WHEN THE GAUGE REGISTERS EMPTY.

AIRSPEED LIMITS (TRUE INDICATED AIRSPEED)

NEVER EXCEED NORMAL OPERATION FLAP OPERATING RANGE MANEUVERING MAX. STRUCTURAL CRUISING SPEED MAX. SPEED FLAPS EXTENDED UTILITY CATEGORY
140 MPH (RED LINE)
115 TO 140 MPH (YELLOW ARC)
50 TO 115 MPH (GREEN ARC)
45 TO 82 MPH (WHITE ARC)
106 MPH
115 MPH
82 MPH

NO INVERTED MANEUVERS PERMITTED.

FLIGHT LOAD FACTORS

HIGH OR LOW ANGLE OF ATTACK - +4.4; INVERTED FLIGHT - 1.76 FLAPS DOWN 43° - +3.5

MAXIMUM WEIGHT - 1500 Pounds.

CENTER OF GRAVITY RANGE

NOTE: DATUM IS FORWARD FACE OF FIREWALL; M.A.C. 58.7°; L.E. MAC+18.8°. Leveling Means - Top Edge of Sheet along fuselage side aft of Door.

25% M.A.C. (+33.5") TO 31% MAC (+37.0")AT 1500 POUNDS AND 23.9% MAC (+32.8") TO 31% MAC (+37.0") AT 1279 POUNDS OR LESS. STRAIGHT LINE VARIATION OF FORWARD LIMITS BETWEEN ABOVE.

NOTES

1. THE FOLLOWING MANEUVERS ARE APPROVED FOR OPERATION IN THE UTILITY CATEGORY WITH RECOMMENDED ENTRY SPEEDS SHOWN:

CHANDELLE - 105; STEEP TURN - 95; STALLS, (EXCEPT WHIP STALLS); LAZY EIGHT - 105; SPINS - USE POWER TO ENGER.

2. MANEUVERS INVOLVING FULL APPLICATION OF FLIGHT CONTROLS SHOULD BE CONFINED TO SPEEDS BELOW MANEUVERING SPEED. ABRUPT MANEUVERS SHOULD BE AVOIDED IN SPEEDS EXCEEDING THE MANEUVERING SPEED.

3. IT IS THE RESPONSIBILITY OF THE PILLOT AND OPERATOR TO SEE THAT THE VEHICLE AND OPERATOR TO SEE THAT THE WEIGHT AND BALANCE ARE WITHIN LIMITATIONS.

THIS AIRPLANE MUST BE OPERATED AS A UTILITY CATEGORY AIRPLANE IN COMPLIANCE WITH THE FLIGHT MANUAL.

NO ACROBATIC MANEUVERS APPROVED EXCEPT THOSE LISTED IN THE FLIGHT BAGGAGE 80 LBS MAX IMEM

PERFORMANCE INFORMATION - LANDPLANE.

THE FOLLOWING OPERATIONAL DATA ARE COMPILED FROM ACTUAL TESTS WITH AIRPLANE AND ENGINE IN GOOD CONDITION AND USING AVERAGE PILOTING TECHNIQUE. ALL PERFORMANCE IS GIVEN FOR 1500 POUNDS GROSS WEIGHT WITH ZERO WIND VELOCITY AND HARD SURFACE LEVEL RUNWAY. TAKE-OFF AND CLIMB PERFORMANCE FIGURES GIVEN BELOW ARE MINIMUM VALUES OBTAINED FOR A WOOD PROPELLER WITH A LOW STATIC RPM (1900). THESE PERFORMANCE FIGURES WILL BE IMPROVED USING A WOOD PROPELLER WITH A HIGHER STATIC RPM OR A MCCAULEY PROPELLER.

STALLING SPEED	CONDITION	Angle of Bank Degrees				
		0°	50°	40°	60°	
MPH T.I.A.S. NO STALL WARNING IS EVIDENT	POWER OFF; FLAPS UP POWER OFF; FLAPS DOWN	53 51	55 52.5	60.5 58	75 72	
THE ABOVE STALLING SPE LOADINGS, THE AIRPLANE	EDS ARE GIVEN FOR MOST F	EARWA	RD CENTER	OF GRAY	ITY.	WITH NORM

	ITEM			ALTITUDE	0°F	0u151 20°F	DE AIR 40°F	TEMPERATUS 60°F.	80°F	1'00°F
PS DOWN	LANDING DIST TO LAND OVER OBSTACLE AT APPROACH SPE ROLL APPROX.	50 Fòo 67 MPH ED. (GRO	TIAS	2000 FT.	1 380 1 470 1560 1650	1430 1520 1610 1700	1480 1570 1660 1750	1530 1620 1710 1800	1580 1670 1760 1850	1625 1720 1810 1 9 00
FLA	TANCE SHOWN.		J -							
APS UP	Take-OFF DI (FEET) TO CL OBSTACLE AT (GROUND RUN DISTANCE SHO	EAR 50 69 MPH APPROX.	TIAS	SEA LEVEL 2000 FT. 4000 FT. 6000 FT.	1440 1750 2120 2650	1570 1900 2320 2920	1700 2050 2540 3220	1830 2210 2760 3570	1960 2390 3020 4100	2100 2580 3330 4750
FL	NORMAL RATE OF CLIMB FEET PER MINUTE	BEST CLIMB SPEED TIAS	77 75 72 70	SEA LEVEL 2000 FT. 4000 FT. 6000 FT.	700 590 480 365	660 550 440 325	620 515 400 2 9 0	585 47 5 36 0 250	550 440 320 210	510 400 280 175

BOTH TAKE-OFF AND LANDING DISTANCES ARE REDUCED APPROXIMATELY 10% FOR EACH 6 MPH WIND VELOCITY.

PERFORMANCE INFORMATION - SKIPLANE

UNDER CAR 4a approval, PERFORMANCE INFORMATION IS NOT REQUIRED.

APPROVED BY: W-. H. Weels
CHIEF, AIRCRAFT BRANCH

DATE: July 11, 1949

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